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# Laminar flow and forced convection heat transfer in plate-type monolith structures by a finite element solution

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**Abstract**—A numerical solution is obtained by the finite element method for the Graetz problem of forced convection heat transfer in the channels of plate-type monolith catalysts. For this purpose a generalized model is constructed for the duct geometry, the fully developed laminar velocity profile is calculated, and the heat transfer problem is solved for both uniform wall heat flux or temperature. The results are summarized in interpolated functional forms providing the axial evolution of the Nusselt number. Such formulae represent a prerequisite to the development of design equations for mass-transfer coefficients in plate-type monolith catalysts for the selective catalytic reduction of  $\text{NO}_x$  with ammonia.

## INTRODUCTION

Forced convection with laminar flow in straight ducts of a constant cross-section has important practical applications. Shah and London [1] have extensively surveyed the well known Graetz problem for different geometries: from circular ducts to parallel plates, from rectangular sections to triangle passages they attempted to systematize thermal boundary conditions, considering those most common in the literature and interesting in applications. They also reported solutions for sine ducts, elliptical forms, circular sectors and other singly connected configurations, that is geometrical shapes delimited by a single closed line. Double and multiple connected ducts were considered, too.

After Shah and London's publication, other authors have addressed the Graetz problem: different numerical methods have been tested on traditional shapes [2-11]; the less conventional geometries have been studied with regard to entrance effects [12-21]; axial conduction has been considered [22-24]; and finally the analysis of non-Newtonian fluids has begun to appear [25-28].

In this article we consider the forced convection heat transfer problem for ducts of complex geometry corresponding to the channels of plate-type monolith catalysts [29, 30]. We first examine the hydrodynamic problem to obtain information on the laminar velocity profiles inside the monolith channels; then we analyze the heat transfer problem, and eventually derive interpolated expressions for the axial evolution of the

Nusselt number with regard to different boundary conditions.

The novelty of this paper lies in the duct geometry considered, as well as in the algorithm adopted for numerical solution of the problem. The choice of the duct geometry is motivated according to two different arguments. First of all, the cross-sections of plate-type monoliths exhibit a peculiar configuration including two non interacting channels. In this paper we illustrate an approach suitable to reduce such a situation to the analysis of a single duct. The same analysis is also expected to provide applied benefits, since catalytic monoliths in plate form are actually of commercial interest for the selective catalytic reduction (SCR) of  $\text{NO}_x$  with ammonia, SCR processes being the most efficient technology available for the denitrification of flue gases from power stations [30]. It has been shown in the literature [31] that gas-solid mass transfer coefficients in monolithic honeycomb SCR reactors with simple channel geometry (circular, square and triangular) are adequately predicted by relying on the similarity with the corresponding heat transfer problems for constant wall temperature or constant heat flux. It is of practical relevance for the modelling of industrial SCR monolith reactors to establish whether the same conclusions apply also to the class of plate-type monolith catalysts: as mentioned above, however, solutions of the thermal problem for this more complex geometry are not available. While the derivation of such solutions on a generalized basis is the scope of the present work, in a future article [32] they will be applied to the specific purpose of developing a design procedure for plate-type monolithic SCR reactors.

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## NOMENCLATURE

$A_H, B_H, C_H, D_H$	interpolation parameters for $Nu(H2)$ —equations (26)–(30)	$v^*$	dimensionless axial velocity—equation (7)
$A_T, B_T, C_T, D_T$	interpolation parameters for $Nu(T)$ —equations (17)–(22)	$\langle v \rangle$	dimensional velocity averaged over the duct cross-section
$c_p$	fluid specific heat	$xyz$	dimensional Cartesian coordinates
$D_h$	hydraulic diameter	$x^*y^*z^*$	dimensionless Cartesian coordinates—equation (7)
$DR$	geometrical parameter representative of cross sectional shape, $= (R - R_0)$	$z_i^*$	Graetz axial coordinate $= z / (Pe_i D_{hi})$ .
$h$	heat transfer coefficient		
$H$	channel pitch		
$H'$	$= 2H$ , see Fig. 1	<b>Greek symbols</b>	
$k$	fluid thermal conductivity	$\Gamma$	cross-sectional boundary
$L$	longitudinal extension of cross section	$\mu$	fluid dynamic viscosity
$L_1$	$= 4.2335 H$ see Fig. 1	$\Theta^*$	dimensionless temperature—equations (4), (5)
$L_2$	$= L - 3.173 H$ , see Fig. 1	$\langle \Theta_{cm}^* \rangle$	dimensionless flow average temperature (cup-mixing)
$Nu$	Nusselt number— $(hD_h/k)$	$\langle \Theta_{\Gamma}^* \rangle$	dimensionless wall average temperature
$p$	pressure	$\rho$	fluid density
$Pe$	Peclet number— $(vD_h \rho c_p/k)$	$\tau$	dimensional peripheral flux
$r$	ratio of dimensionless areas of (B) and (A) sections, respectively— $(S_B D_{hB}^2 / S_A D_{hA}^2)$	$\langle \tau \rangle$	dimensional peripheral average flux
$r_D$	ratio of hydraulic diameters of (B) and (A) sections, respectively— $(D_{hA}/D_{hB})$	$\tau^*$	dimensionless peripheral flux
$r_v$	ratio of dimensional average velocities of sections B and A, respectively— $(\langle v \rangle_A / \langle v \rangle_B)$	$\phi$	characteristic angle of the cross section ( $= 57^\circ$ , see Fig. 1).
$Q$	volumetric flow rate		
$R$	geometrical parameter representative of cross sectional shape— $(L/2H)$	<b>Superscripts</b>	
$R_0$	limiting value of $R$ ( $= 2.116$ , see Fig. 1)	*	dimensionless variable
$S$	cross-sectional area	$\infty$	asymptotic value for $z \rightarrow \infty$ .
$T$	dimensional temperature		
$T_0$	value of the initially uniform temperature profile	<b>Subscripts</b>	
$\langle T_{cm} \rangle$	dimensional flow average temperature (cup-mixing)	A	relative to (A) geometry—Fig. 1
$\langle T_{\Gamma} \rangle$	dimensional wall average temperature	AB	relative to (A) + (B) geometry, i.e. to elementary monolith cells—Fig. 1
$v$	dimensional axial velocity	B	relative to (B) geometry—Fig. 1
		cm	cup-mixing average
		H2	relative to (H2)-condition
		T	relative to (T)-condition
		$\Gamma$	calculated at the duct wall.

Concerning the numerical solution of the partial differential equations associated with either the hydrodynamic or the thermal problem, in this work we adopt the Galerkin finite element method (FEM) rather than more usual approaches based on finite differences or orthogonal collocations. Indeed, the finite element analysis offers some advantages over other methods. Primarily, it gives powerful tools, namely irregular meshes, isoparametric elements [33] and graphical gridding, to reproduce complex geometries such as those involved in the present work; also, FEM provides local and global error estimates

useful in adaptive refinement processes and convergence studies [34–36].

## THEORY AND METHODS

*Geometrical analysis*

For our purposes it is first necessary to identify the geometrical domain of interest. In the following, a geometrical reference model is presented.

Inspection of commercial monolith plate-type catalyst matrices shows the existence of an elementary cell in the monolithic section. As represented in Fig. 1, the

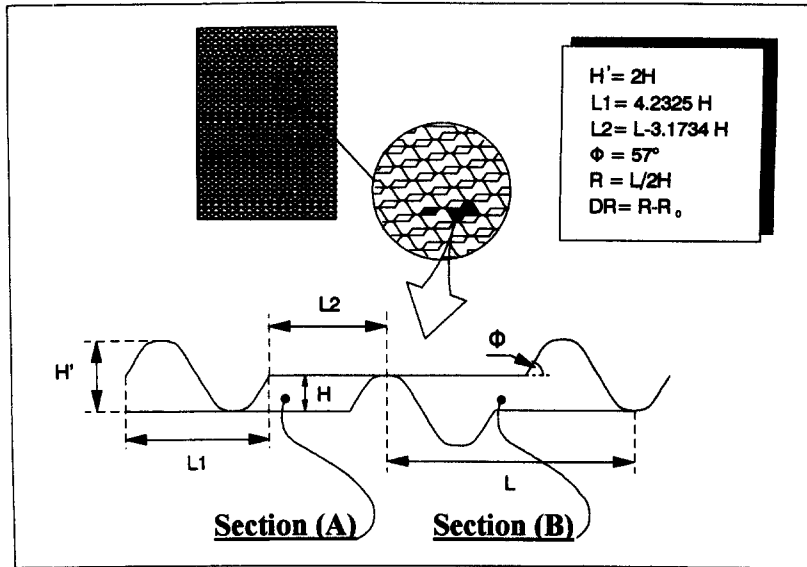


Fig. 1. Typical configuration of plate-type monolith catalyst and two-channel parametric geometrical model. All lengths are in cm.

modular monolith matrix includes two different kinds of geometries which are alternated side by side: the elementary cell consists of a section A, similar to a parallelogram, and a section B, which exhibits characteristic sinusoidal appendices serving as spacers. A representative geometrical parameter  $DR$  is defined, which equals the difference between the ratio,  $R (= L/2H)$  and its limiting value,  $R_0 (= 2.116)$ , corresponding to compenetration of the sinusoidal appendices with the disappearance of Section A (Fig. 1).

Six values of  $DR$  have been chosen for the numerical study of fluid dynamics and heat transfer in the forced convection flow in such sections (see the second column of Table 1). The six values are selected to cover the field of possible industrial interest (a representation of such geometries appears in Fig. 2). Though in the following they are studied individually with respect to both the hydrodynamic and the heat transfer problem, it is worth noting that the six configurations actually serve as a discrete representation of the general geometrical model of plate-type monoliths.

Finally, it is necessary to specify that even if under

normal operating conditions communication between the channels may be possible, however the absence of every form of interaction is assumed in the following analysis. This hypothesis results in the identification of two different and separated cross-sections A and B with different geometries, each of them requiring a dedicated investigation of fluid dynamics and heat transfer. Thus, for each  $DR$  value the analysis of such two domains will be carried out, as well as a subsequent combination of results to arrive at a unitary representation referring to a single pseudo cross-section of the monolith channel.

#### The velocity problem

We consider steady-state flow of a fluid with a fully developed laminar velocity profile. Assuming constant physical properties  $\rho$ ,  $\mu$  for the fluid and neglecting body forces, centrifugal effects, Coriolis action and electromagnetic interaction, the momentum equation can be written in the following dimensional form [1]:

$$\nabla^2 v = \frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} = \frac{1}{\mu} \frac{dp}{dz} = c_1, \quad (1)$$

where  $v$  is the axial component of velocity which must satisfy the boundary condition at the border

$$v \equiv 0 \quad \forall (x, y) \in \Gamma. \quad (2)$$

We do not take advantage of the oblique symmetry of geometries to avoid the presence of mixed derivatives in equation (1). So we solve the velocity problem in the form of equations (1) and (2), identifying the mathematical boundary  $\Gamma$  with the real channel walls of the monolith. Then we choose a dimensional formulation to impose the same pressure gradients in

Table 1. Geometric and fluid dynamic properties of the investigated geometries

Sample	$DR$	$r$	$r_D$	$r_v$
1	0.779	0.55777	0.68729	0.39702
2	1.837	0.63827	0.75701	0.51689
3	3.689	0.72034	0.84897	0.63244
4	5.532	0.80671	0.88427	0.69588
5	7.052	0.81096	0.90461	0.75363
6	11.628	0.87216	0.93667	0.85396

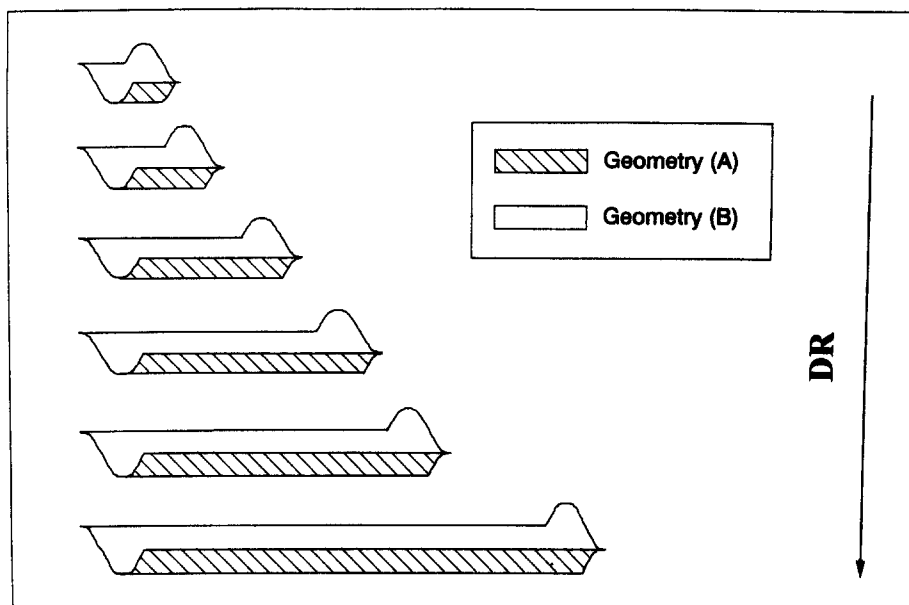


Fig. 2. Elementary cell of the assumed plate-type monolith geometry.  $DR$  increases from top to bottom.

channels A and B, and we assume the constant  $c_1$  is equal to unity.

Notably, the assumption of laminar flow is consistent with the flow conditions prevailing in SCR monolith catalysts.

#### The temperature problem

This problem concerns the solution of the energy conservation equation for a fluid in laminar flow in a duct under different thermal boundary conditions [1]. The fluid properties  $\rho$ ,  $k$ ,  $c_p$  are assumed constant and the laminar fully developed velocity profile is known by assumption, as determined from solution of the hydrodynamic problem. Viscous effects and axial thermal diffusion are neglected and we exclude phase transitions and chemical energy changes. Accordingly, the governing equation is [1]

$$v\rho c_p \frac{\partial T}{\partial z} = k \left( \frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} \right) \quad (3)$$

with the temperature,  $T$ , being the only unknown.

Three different boundary conditions are most often considered [1]: the ( $T$ )-condition involves a uniform and constant temperature profile at the wall, both peripherally and axially; the ( $H2$ )-condition requires a uniformly distributed heat flux on the channel section and a constant average heat flow in the axial direction; the ( $H1$ )-condition imposes a uniform wall temperature at any cross-section and a constant wall heat transfer rate in the axial direction. In view of the eventual application of the present results to modelling of SCR reactors, in this paper only the ( $T$ ) and ( $H2$ ) conditions are examined. In fact, they exhibit the same mathematical formulation of the analogous

problem concerning gas–solid mass transfer in the monolith channels in the two limiting cases of infinitely fast kinetics (purely physical regime) and infinitely slow kinetics (purely chemical regime), respectively [37–39]. Accordingly, the Nusselt numbers derived from solutions of the two thermal problems provide also the limiting values of the dimensionless gas–solid mass transfer coefficients (Sherwood numbers) for fast and slow kinetics, respectively [31]. This point will be further developed in a forthcoming paper [32].

Referring to the ( $T$ ) and ( $H2$ ) conditions, the following dimensionless temperatures are usually introduced [1]:

$$\Theta^* = \frac{T - \langle T_T \rangle}{T_0 - \langle T_T \rangle} \quad \text{for } (T)\text{-condition} \quad (4)$$

$$\Theta^* = \frac{T - T_0}{\frac{\tau D_h}{k}} \quad \text{for } (H2)\text{-condition.} \quad (5)$$

Thus, equation (3) can be rewritten in dimensionless form as

$$v^* \frac{\partial \Theta^*}{\partial z^*} = \frac{\partial^2 \Theta^*}{\partial x^{*2}} + \frac{\partial^2 \Theta^*}{\partial y^{*2}}, \quad (6)$$

where

$$x^* = \frac{x}{D_h} \quad y^* = \frac{y}{D_h} \quad z^* = \frac{z}{Pe D_h} \quad v^* = \frac{v}{\langle v \rangle}. \quad (7)$$

Finally, in both situations the initial condition involves a uniform thermal profile, at the value of

unity for the (*T*) case and zero for the (*H2*) case. Then we have

$$(T)\text{-condition} \begin{cases} \text{B.C. } \Theta^* = 0 \quad \forall (x^*, y^*, z^*) \in \Gamma \\ \text{I.C. } \Theta_0^* = 1 \quad \forall (x^*, y^*) \text{ at } z^* = 0. \end{cases} \quad (8)$$

$$(H2)\text{-condition} \begin{cases} \text{B.C. } \tau^* = 1 \quad \forall (x^*, y^*, z^*) \in \Gamma \\ \text{I.C. } \Theta_0^* = 0 \quad \forall (x^*, y^*) \text{ at } z^* = 0. \end{cases} \quad (9)$$

### Problem analysis

The study of fluid dynamics and heat transfer in the geometrical model of the plate structure is developed along four stages:

(1) Solution of the hydrodynamic problem, equations (1) and (2); since the monolith configuration includes two different kinds of channel sections, the velocity problem has to be solved separately for domains A and B. The operation is repeated for the six different values of the parameter *DR*.

(2) Definition of a single representative fluid dynamic variable: the construction of a global average velocity ( $\langle v \rangle_{AB}$ ) is proposed. The operation involves the combination of the average velocities in channels A and B ( $\langle v \rangle_A, \langle v \rangle_B$ ) for the six different values of *DR* and the subsequent definition of the relation between  $\langle v \rangle_{AB}$  and *DR*. This results in the construction of a functional from  $\langle v \rangle_{AB}(DR)$  which allows a global one-dimensional representation of fluid dynamics for every value of *DR*. Construction of  $\langle v \rangle_{AB}$  is described in the Results and Discussion Section.

(3) Solution of the temperature problem, equations (3) and (4) (*T*-condition) or equations (3)–(5) (*H2*-condition), and determination of the streamwise evolution of the Nusselt number along the axial coordinate. As in point 2 above, the study of two different Sections A and B is required for six different values of *DR*, so that 12 discrete axial profiles of the Nusselt number are generated.

(4) Definition of a single representative heat transfer variable: like in the velocity problem the construction of a single Nusselt number ( $Nu_{AB}$ ) is proposed for the six geometries to represent the heat transfer process in a global way. The operation is organized in three subsequent steps:

(a) combination of channel A and B Nusselt numbers ( $Nu_A, Nu_B$ ) and definition of a global Nusselt number ( $Nu_{AB}$ )

$$Nu_{AB} = \frac{D_{hAB} h_{AB}}{k} = \frac{D_{hAB} \langle \tau \rangle_{AB}}{k(\langle T_{cm} \rangle_{AB} - \langle T_T \rangle_{AB})}. \quad (10)$$

The calculations are made using the discrete axial profiles of  $Nu_A$  and  $Nu_B$  constructed by numerical solution of the temperature problem and yields six discrete values of global  $Nu_{T,AB}$  as well as six values

of global  $Nu_{H2,AB}$  along  $z_{AB}^*$  the global axial coordinate.

(b) Empirical fitting of the relationship between  $Nu_{AB}$  and  $z_{AB}^*$ . This operation is carried out for every chosen value of *DR*, and leads to the definition of six global  $Nu_{T,AB}$  and six global  $Nu_{H2,AB}$  continuously defined along  $z_{AB}^*$ .

(c) Identification of analytical models representative of the heat transfer characteristics of all the considered geometries, reproducing in a continuous fashion the relationship between the axial evolution of the global Nusselt number and the parameter *DR*: the results obtained from FEM simulations are extended to the whole geometrical model and a generalized functional form  $Nu_{AB}(z_{AB}^*, DR)$  is derived for both (*T*) and (*H2*) conditions.

### Numerical methods

The analysis of the problem requires the solution of partial differential equations as well as the regression of results (in terms of Nusselt numbers) as functions of  $z_{AB}^*$  and *DR*.

Nonlinear regressions are carried out using the FORTRAN program BURENL [40], implementing a collection of direct and indirect search methods for minimization of the residual sum of squares.

For the solution of PDEs, an adequate numerical approach must be implemented in order to meet satisfactorily the boundary conditions associated with the complex duct geometry. In this work we apply the Galerkin finite element method [33].

The basic idea of FEM consists in reducing the problem size by dividing the domain into smaller regions, called elements, where the governing equations are approximated according to the methods of weighted residuals. Within each element the simplified governing equations have the same formal expression, and can be assembled to retain the unity of the problem and the continuity of the solution to generate an overall system of algebraic or ordinary differential equations, depending on the nature of the original PDEs. Since the element is an independent unit, it can be locally adjusted to describe a particular boundary: this secures a higher accuracy for the description of complex domains than offered by, e.g. finite differences of orthogonal collocations.

In order to allow an easy description of the geometrical model of Fig 1, isoparametric serendipity second-order elements (triangular-six nodes and quadrilateral-nine nodes) [33] and graphical gridding have also been implemented in a specific FORTRAN code. Also, solution of the overall system of equations has been carried out by Gaussian elimination (for axial-developed problems) and by Gear's multivalued ODE integration method [41] (for axially developing problems), achieving efficient convergence. Finally, an accurate phase of postprocessing produces graphical tools to check the results qualitatively and quantitatively.

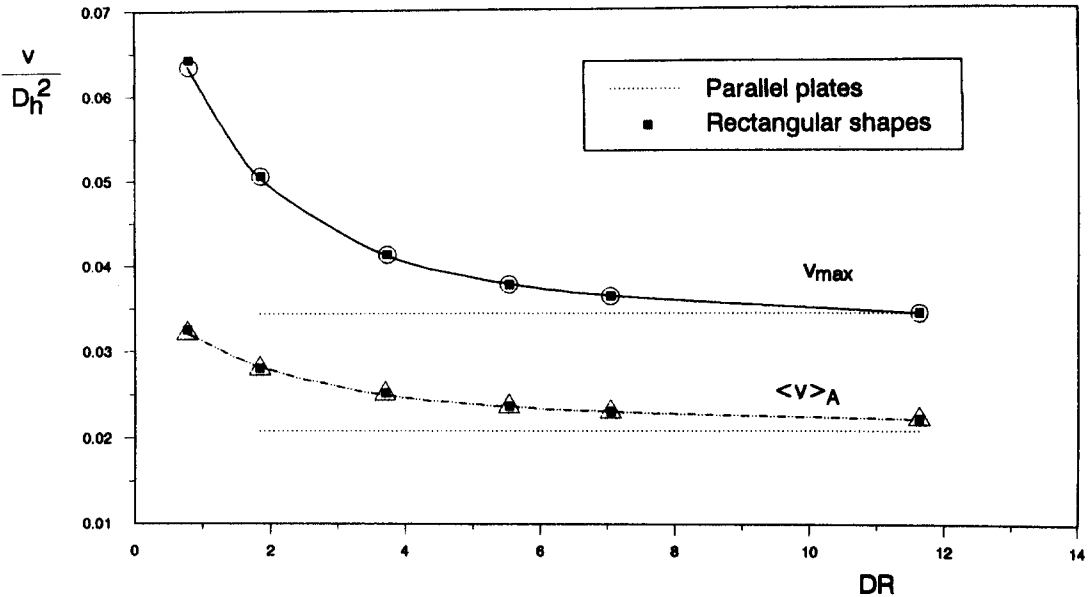


Fig. 3. Velocity problem: maximum and average velocity for channels A vs  $DR$ . The velocities are divided by the squared hydraulic diameter so as to be compared with rectangular shape and parallel plate literature data [1]. All results are relative to a unit  $c_1$  constant [see equation (1)].

## RESULTS AND DISCUSSION

### The velocity problem

*Section (A).* The study of channel A reveals a similarity with the hydrodynamics of laminar flow in rectangular ducts. Indeed, large aspect ratios make the influence of the oblique sides negligible, so that a parabolic velocity profile covers most of each section, giving average and maximum velocities close to the corresponding literature data [1] for rectangular channel shapes (Fig. 3). In the limit of  $DR \rightarrow \infty$  the situation of parallel plates is recovered.

*Section (B).* The analysis of laminar flow in channels B shows a different and more complex situation, as illustrated in Fig. 4. For large values of  $DR$  the velocity profile includes two different contributions: on one hand the curvilinear appendices exhibit a typical concentric sinusoidal profile; on the other hand the central portion shows an approximately parabolic profile quite similar to the case of long rectangles or parallel plates. For small values of  $DR$  the two contributions compenetrates each other and it becomes impossible to trace their distinctive features separately.

A synthetic representation of average and maximum velocities upon varying  $DR$  is shown in Fig. 5: even if increasing  $DR$  reduces the average velocity, the maximum velocity does not converge to that characteristic of parallel plates. Thus, differently from Sections A, the parallel plates provide an asymptotic situation for B geometries only in a mean sense.

*Combination.* In order to achieve a unitary representation of fluid dynamics for the two-channel

model of Fig. 1 we define a global average velocity ( $\langle v \rangle_{AB}$ ). For each considered value of  $DR$  it is calculated according to the following equation:

$$\langle v \rangle_{AB} = \frac{\langle v \rangle_A S_A + \langle v \rangle_B S_B}{S_A + S_B} \quad (11)$$

The volumetric flow rate ( $Q$ ) is treated in the same way, too.

To generalize the results, both  $\langle v \rangle_{AB}$  and  $Q$  are then expressed as functions of the geometrical parameter  $DR$ . The resulting empirical formulas are shown below [equations (12) and (13)]

$$\langle v \rangle_{AB} = 0.0154 + 9.78 \times 10^{-3} DR^{-2.44 \times 10^{-1}} \times \exp(-1.48 \times 10^{-1} DR) \quad (12)$$

$$Q_{AB} = 4.986 \times 10^{-2} + 1.035 \times 10^{-2} DR \quad (13)$$

Both equations yield less than 0.3% errors when compared to calculated data.

### The temperature problem

In the following, the (T) and (H2) boundary conditions will be treated separately for the sake of clarity.

#### (T)-Condition

*Section (A).* For channel A the use of about 300 nodes and nonuniform meshes leads to converged results in accordance with literature data [1] for rectangular duct shapes, which confirms the negligible contribution of oblique sides. The results plotted in Fig. 6 show that increasing  $DR$  brings about greater

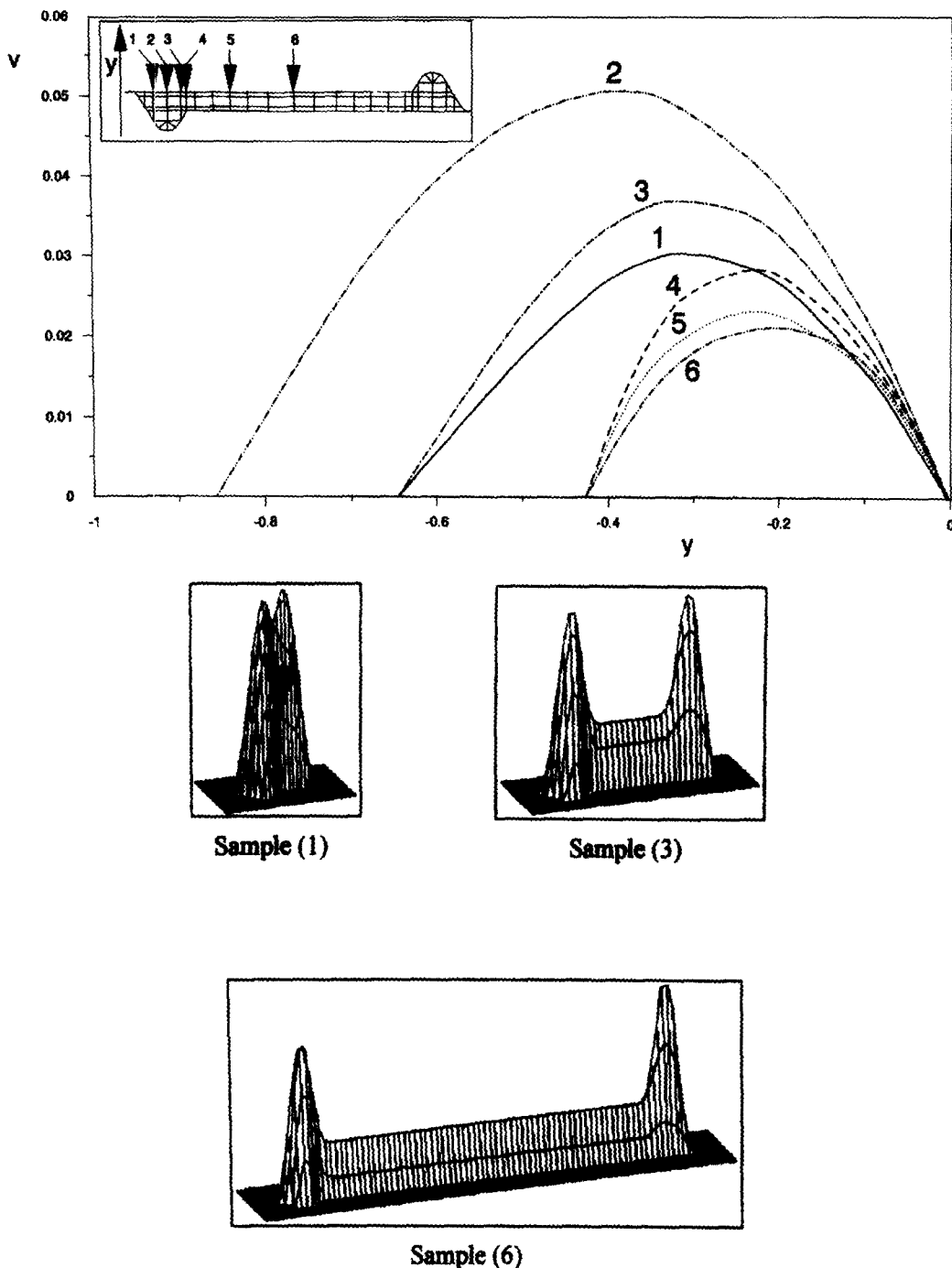


Fig. 4. Velocity problem: fully developed laminar velocity profiles in channels B.

Nusselt numbers: in the limit of  $DR \rightarrow \infty$  the same  $Nu$  evolution as in parallel plates is recovered.

*Section (B).* The study of Section B reveals a more complex situation dominated by two different asymptotic behaviours (see Fig. 7):

(1) In the entrance region (see Table 2 and Fig. 8) the smaller thermal resistance of the parallel sides in section B prevails on the longer effective conduction

path length for energy transfer from wall to bulk associated with the corner-shaped sinusoids, so that heat transfer is controlled by the central portion of Section B, and a thermal behavior similar to that of ducts with rectangular cross-sections is evident. Also, increasing  $DR$  emphasizes the importance of the central zone, resulting in a reduced average heat transfer resistance. Thus, like for Section A, at small  $z_0^*$  increasing  $DR$  results in greater Nusselt numbers: in the limit

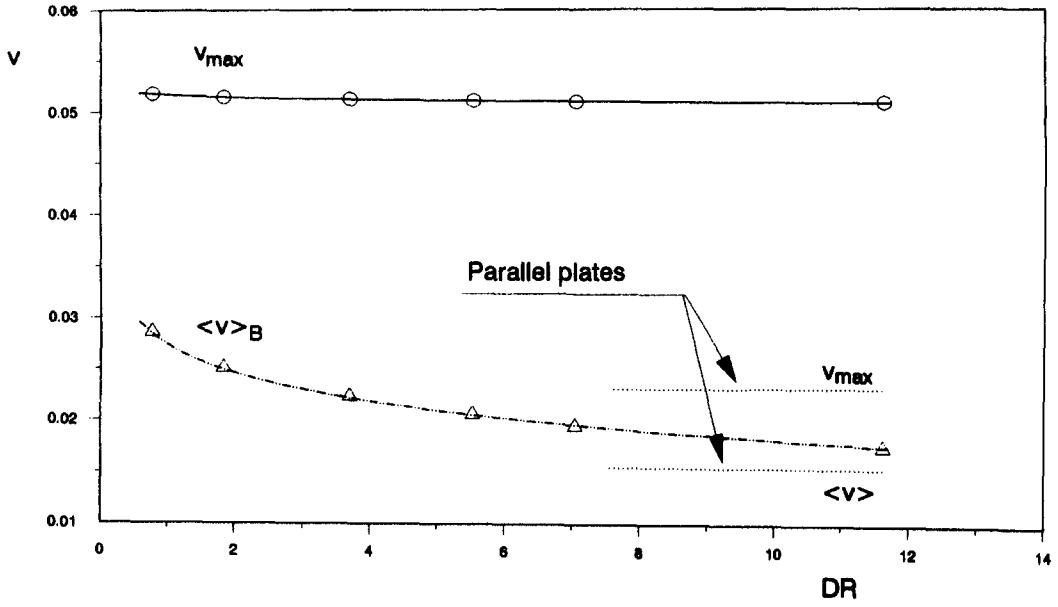


Fig. 5. Velocity problem: maximum and average velocity for channels B vs  $DR$ . All results are relative to a unit  $c_1$  constant [see equation (1)].

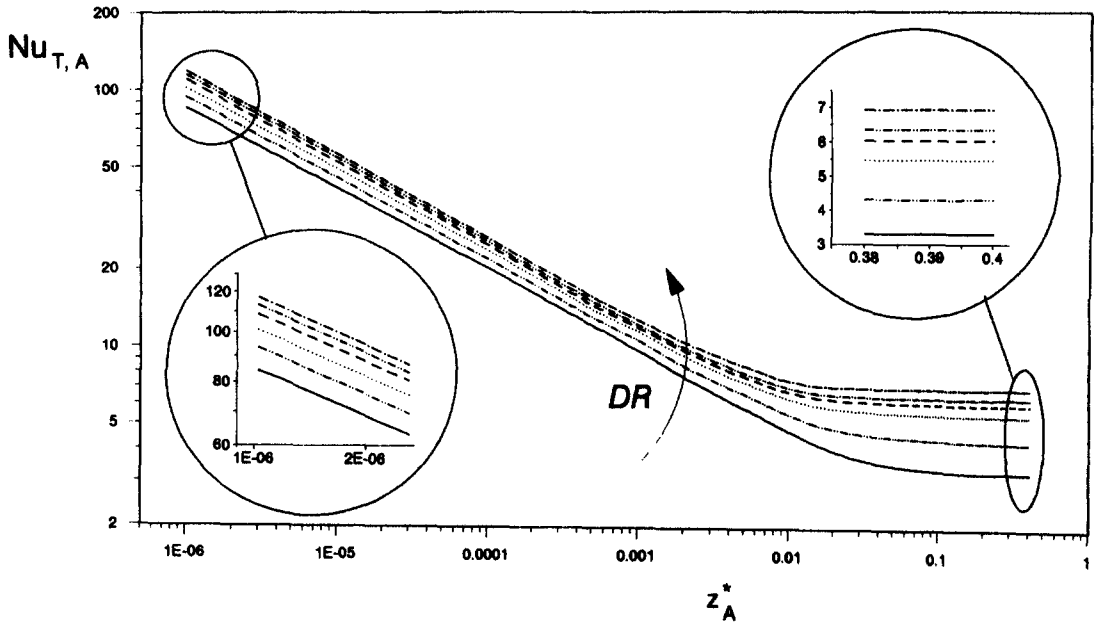


Fig. 6. (T)-condition: axial evolution of the Nusselt number of channels A.

of  $DR \rightarrow \infty$ , the  $Nu$ -evolution typical of parallel plates is recovered. In fact, for the sixth geometry ( $DR = 11.628$ ) at  $z_B^* = 1.0 \times 10^{-6}$  we calculate  $Nu = 118.19$ , which is close to literature data [1] for parallel plates (122.94).

(2) On the other hand, in the region of nearly developed thermal profile the Nusselt number of Section B is controlled by the curved appendices (see Table 2 and Fig. 8). Because of its smaller thermal

resistance, in fact, the central zone is reached before thermal equilibrium with the wall, so that its contribution to heat transfer becomes already negligible when the T-profiles in the sinusoids are still evolving. Under such conditions an increment of  $DR$  produces two opposite effects: while it enhances heat transport in the central region, it also causes the same central portion to reach earlier thermal equilibrium with the wall, which makes the sinusoidal appendices con-



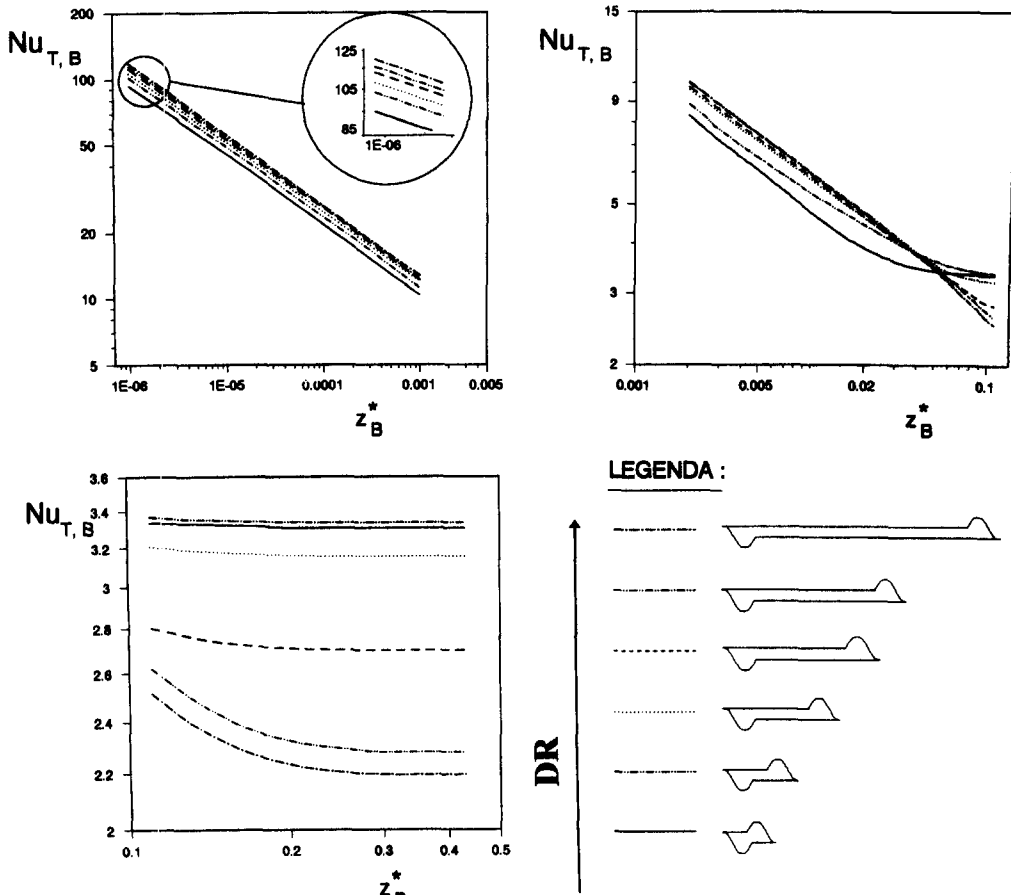


Fig. 7. (*T*-condition: axial evolution of the Nusselt number for channels B. For the sake of clarity the representation is subdivided into three different portions showing the rectangle-like and sine-like asymptotic behaviours and the intermediate transition.

trolling the wall—bulk heat transfer process over a longer stretch of the axial coordinate. For small *DR* the first effect is stronger; for larger sections the second one dominates: this explains the dependence of the asymptotic Nusselt number on *DR* characterized by an inflection point shown in Figure 9. Notice that, in contrast with the situation of the entrance region, here for *DR* → ∞ one approaches asymptotically the heat transfer behaviour of the sinusoidal section: our numerical solution yields  $Nu_B^\infty = 2.195$  (confirmed by an additional test for *DR* = 15), which is close to the literature data of 2.12 for sinusoidal ducts [1]; the small difference is due to an interaction with the central region which cuts off at three quarters the sinusoids and slightly enhances the heat transfer efficiency. The weak maximum of  $Nu_B^\infty$  in the region of small *DR* in Fig. 9 can be rationalized by the transition to a completely different fluid dynamic situation devoid of the central parabolic contribution and exclusively dominated by the sinusoidal appendices.

*Combination.* In order to achieve a unique Nusselt number of a one-dimensional representation, a global  $Nu_{T,AB}$  is calculated, using the discrete axial profiles of  $Nu_{T,A}$  and  $Nu_{T,B}$  generated by numerical solution of

the temperature problem. The formulae employed for this purpose involve the fluid dynamic hypothesis of equal pressure loss in the channels. They make use of the dimensionless relationships (4), (5) and (7) and of the Nusselt number definition [equation (10)]: on this subject notice that Sections A and B have different hydraulic diameters  $D_h$  and that the construction of  $Nu_{T,AB}$  requires the use of a combined hydraulic diameter ( $D_{h,AB}$ ). Finally the formulae consider the phase displacement between the Graetz-coordinates of the channels ( $z_A^*$  and  $z_B^*$ ) and that of the elementary cell ( $z_{AB}^*$ )

$$z_A^* = \frac{z}{Pe_A D_{hA}} = z_{AB}^* \frac{(1+r_D^2 r)(1+r_v r_D^2 r)}{r_v r_D^2 (1+r_D r)^2} \quad (14)$$

$$z_B^* = \frac{z}{Pe_B D_{hB}} = z_{AB}^* \frac{(1+r_D^2 r)(1+r_v r_D^2 r)}{(1+r_D r)^2}, \quad (15)$$

where  $r_D$  equals the ratio of hydraulic diameters ( $D_{h,A}/D_{h,B}$ );  $r_v$  compares the dimensional average velocities ( $\langle v \rangle_A / \langle v \rangle_B$ ); and  $r$  represents the proportion between dimensionless areas ( $S_A D_{hB}^2 / S_B D_{hA}^2$ ); for

Table 2. Entrance and asymptotic Nusselt numbers for the (T)-condition in channel B

$z_B^*$	Sample (1)	Sample (2)	Sample (3)	Sample (4)	Sample (5)	Sample (6)
$0.10 \times 10^{-6}$	$0.93461 \times 10^2$	$0.10170 \times 10^3$	$0.10645 \times 10^3$	$0.11137 \times 10^3$	$0.11450 \times 10^3$	$0.11820 \times 10^3$
$\infty$	$0.33081 \times 10^1$	$0.33400 \times 10^1$	$0.31550 \times 10^1$	$0.26990 \times 10^1$	$0.22810 \times 10^1$	$0.21950 \times 10^1$

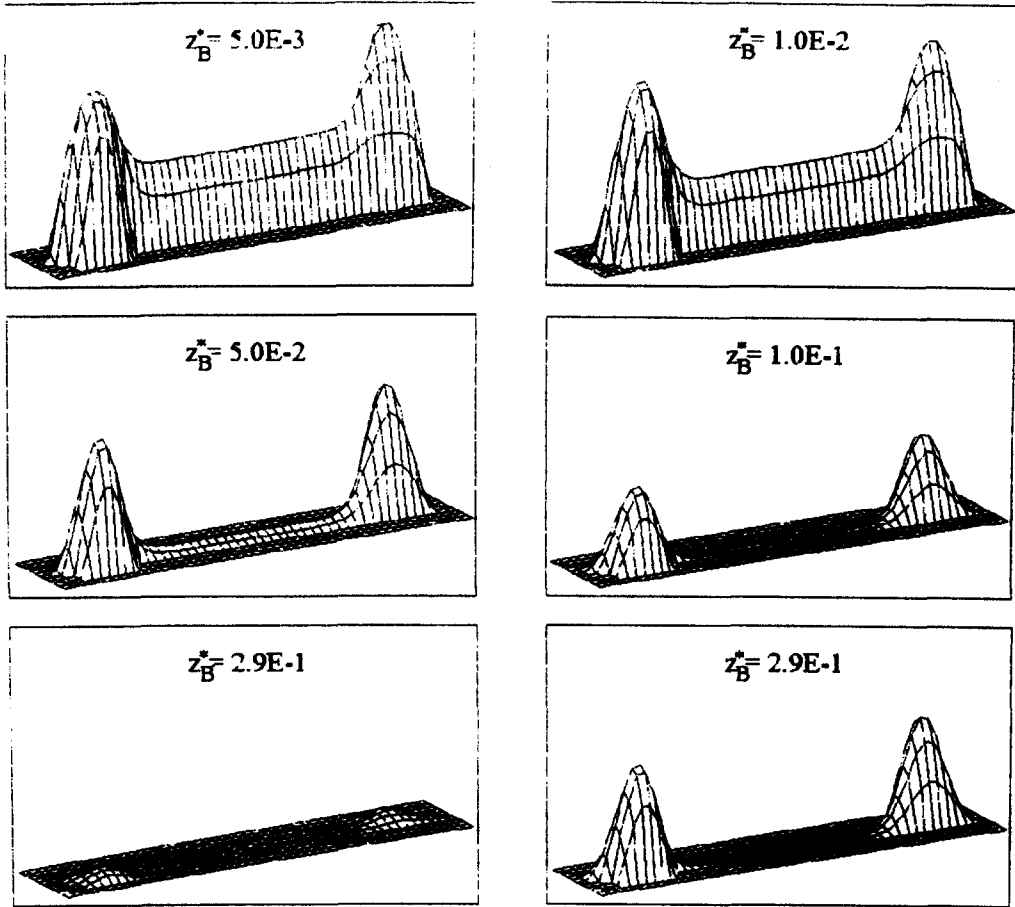


Fig. 8. (T)-condition, sample 4, channel B: temperature distribution over the cross-section at different values of the axial coordinate.

numerical values of these parameters see the last columns of Table 1.

So for the (T)-condition we have

$$Nu_{T,Ab} = \frac{(1+r_D^2r) \langle \Theta_{cm}^* \rangle_B Nu_B + \langle \Theta_{cm}^* \rangle_A Nu_A r}{(1+r_D r) \left( \frac{\langle \Theta_{cm}^* \rangle_B + \langle \Theta_{cm}^* \rangle_A r v_D^2 r}{1+r v_D^2 r} \right)} \quad (16)$$

The calculation, made for every chosen value of DR, produces the axial profiles shown in Fig. 10. Here, the calculated results are represented by triangular symbols, whereas, a solid line shows the inter-

polating model based on the following functional form:

$$Nu_{T,AB} = Nu_{T,AB}^\infty + A_T (1000z_{AB}^*)^{-B_T} \times \frac{\exp(-C_T(D_T + z_{AB}^*))}{\exp(-C_T(D_T + z_{AB}^*)) + \exp(C_T(D_T + z_{AB}^*))} \quad (17)$$

The form of equation (17) derives both from empirical consideration and from usual literature expressions [42]. Among the tested models, it best reproduces the characteristic shape due to the presence of two different asymptotic Nusselt numbers ( $Nu_A^\infty$  and  $Nu_B^\infty$ ): it secures less than 0.5% errors for all the six values of DR.

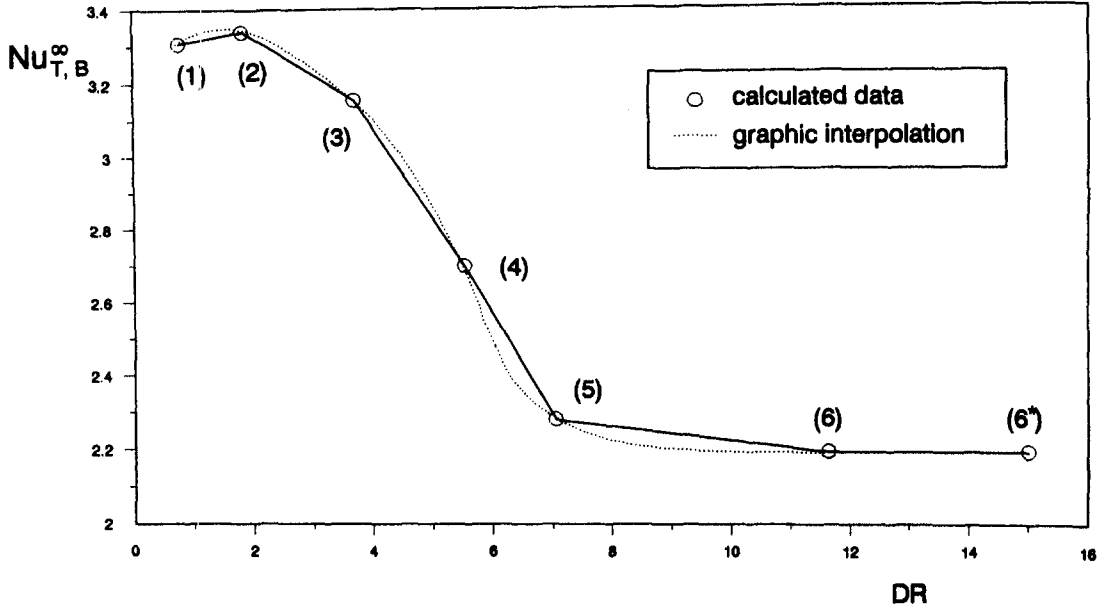


Fig. 9. (*T*)-condition: asymptotic Nusselt number in channels B vs *DR*. In addition to the values calculated for the six samples of Table 1 the asymptotic number of  $DR = 15$  is reported to confirm the trend for  $DR \rightarrow \infty$ .

To extend the results to all the other possible configurations of the geometrical model, the calculated values of  $A_T$ ,  $B_T$ ,  $C_T$ ,  $D_T$  and the combined  $Nu_{T,AB}^{\infty}$  are fitted as functions of  $DR$ . The results of this analysis are summarized in the following:

$$A_T(DR) = 11.665 + 192.167DR^{0.2851} \exp(-0.9027DR) \\ \times \exp(-1.5DR^2 + 1.267DR + 8.8671) \quad (18)$$

$$B_T(DR) = 0.3467 + 5.0373 \times 10^{-2} DR^{0.30723} \\ \times \exp(-0.49634DR) \quad (19)$$

$$C_T(DR) = 10.1252 + 0.9877 \exp(0.1186DR) \quad (20)$$

$$D_T(DR) = -0.09948 + 3.5016DR^{0.76301} \\ \times \exp(-2.2214DR^{0.53113}) \quad (21)$$

$$Nu_{T,AB}^{\infty}(DR) = 2.195 - 0.03524(DR^{4.2495} + 6.9217DR) \\ \times \exp(-0.7039DR) \log(0.1899DR). \quad (22)$$

Notice that all the functional forms above derive from purely empirical considerations: since we are performing a regression on regression results, it is not possible to associate any physical meaning with the parameters values. In terms of accuracy, however, the above formulation seems quite satisfactory: it reproduces the relation  $Nu_{T,AB}(DR, z_{AB}^*)$ , and by means of a simple one-dimensional energy balance integration

$$\frac{d\langle \Theta_{cm}^* \rangle}{dz^*} = -4Nu\langle \Theta_{cm}^* \rangle - \langle \Theta_{\pm}^* \rangle \quad (23)$$

it gives the axial evolution of the flow average global

temperature ( $\langle \Theta_{cm}^* \rangle_{AB}$ ) with less than 5% errors when compared to FEM rigorous but extremely more expensive results.

#### (H2)-Condition

*Section (A).* The axial evolution of the calculated Nusselt number upon varying  $DR$  are shown in Fig. 11. A section of larger  $DR$  gives a bigger  $Nu_B$ , although the growth of the Nusselt number is slower than in the (*T*) case. Similar to that case, however, the behaviour of typical of parallel plates is asymptotically recovered for  $DR \rightarrow \infty$ .

*Section (B).* For the (*H2*)-condition all numerical studies show a different situation with respect to the (*T*)-condition (Fig. 12). Indeed, the reason for the distinction is clear: the (*H2*)-condition, involving a constant heat flux, indefinitely allows for heat exchange also in the central block of the channel section. Therefore, the present condition lets the behaviour of parallel plates be the natural asymptote ( $DR \rightarrow \infty$ ) for the (*B*) geometry at every  $z_B^*$ .

*Combination.* The global Nusselt number ( $Nu_{H2,AB}$ ) is calculated according to the following equation:

$$Nu_{H2,AB} = \frac{1}{\frac{Nu_B r_D^2 r + Nu_A}{Nu_A Nu_B (1 + r_D^2)} = 4z_{AB}^* \frac{r(r_D - 1)^2}{r_D(r_D + 1)^2}} \quad (24)$$

Equation (24) derives from a combination process similar to that for the (*T*)-condition: the parameters  $r$ ,  $r_v$  and  $r_D$  have the same meaning (Table 1) and  $z_{AB}^*$  still represents the Graetz-coordinate based on a glo-

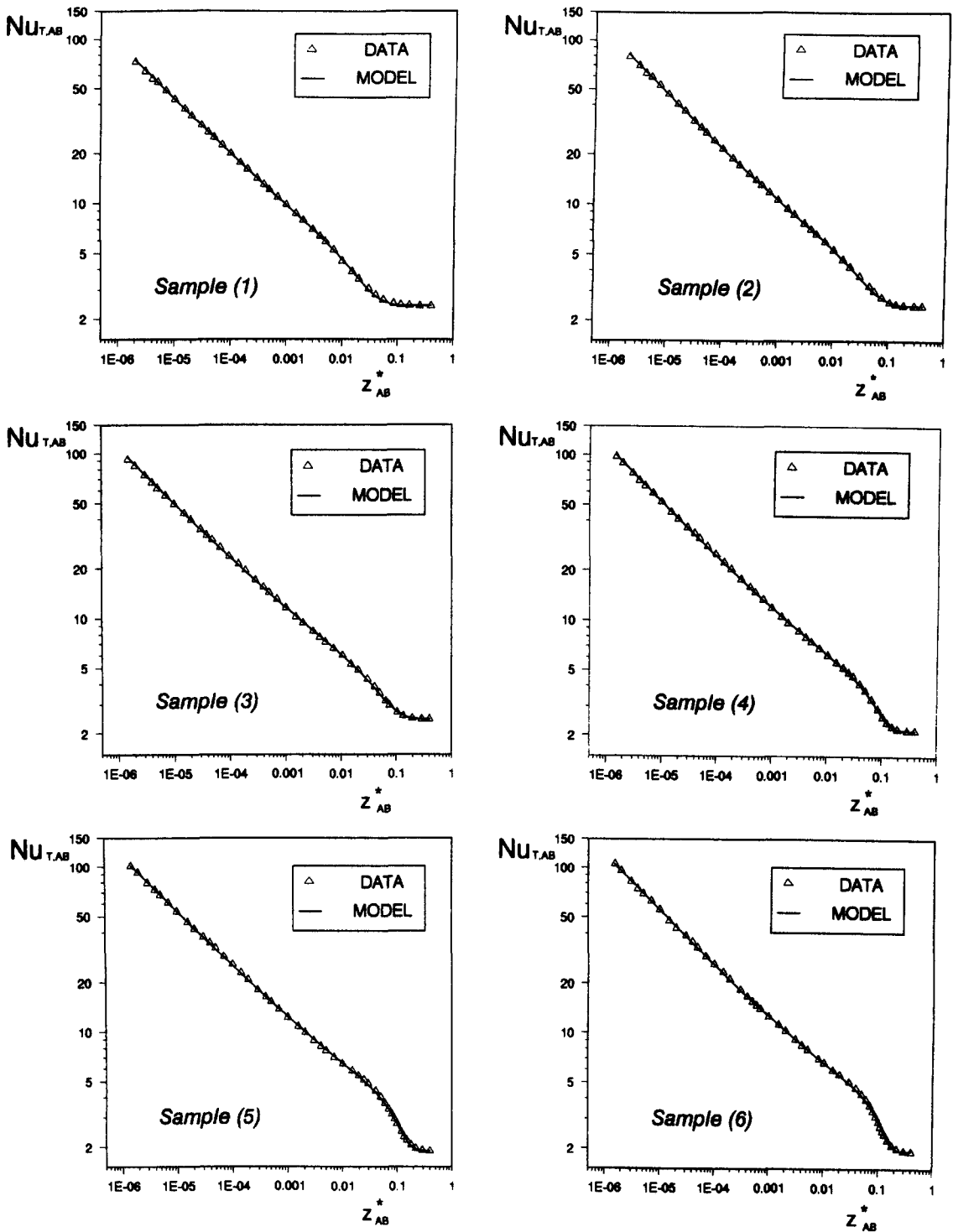


Fig. 10. (T)-condition: axial evolution of the global Nusselt number. Discrete results and interpolating model.

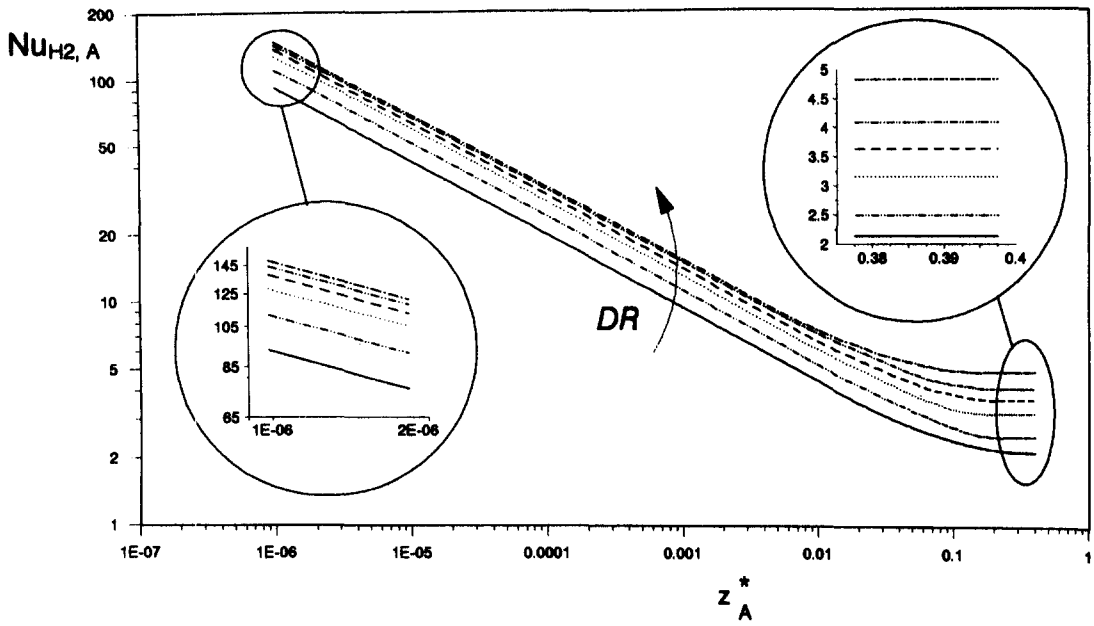


Fig. 11. (H2)-condition: axial evolution of Nusselt number in channels A. Legend of the curves as in Fig. 7.

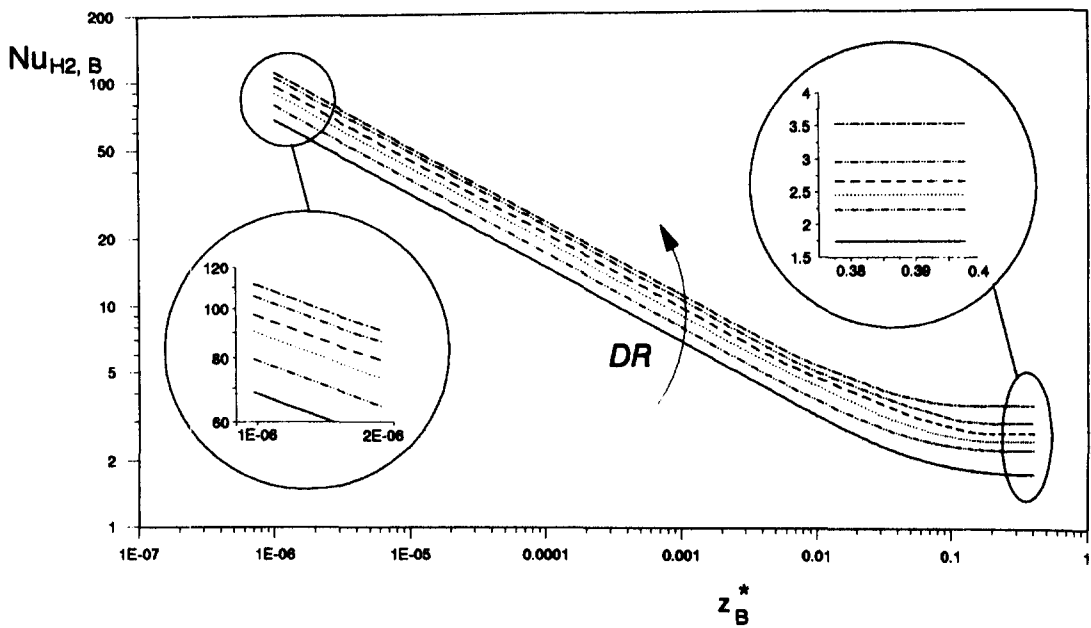


Fig. 12. (H2)-condition: axial evolution of Nusselt number in channels B. Legend of the curves as in Fig. 7.

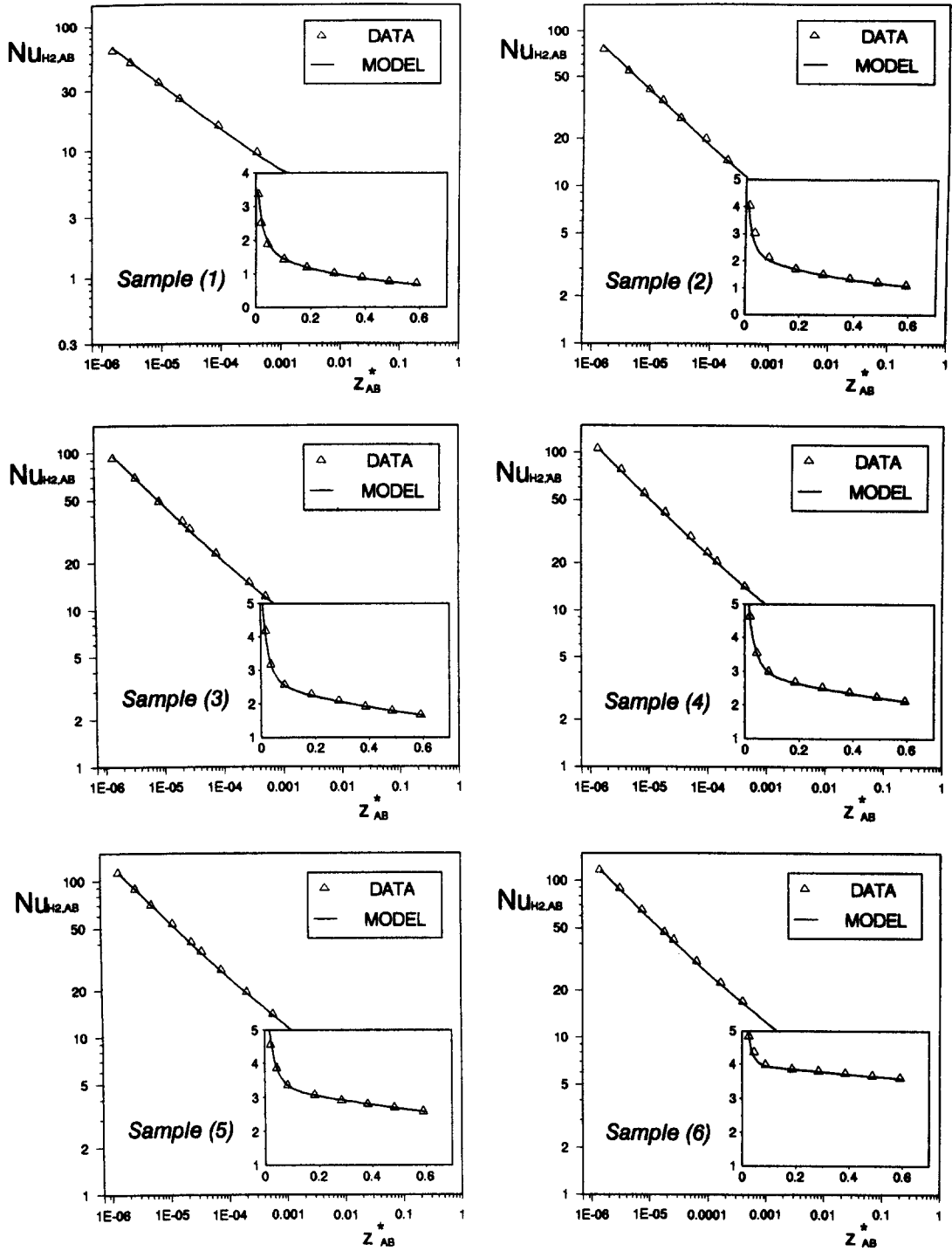


Fig. 13. (H2)-condition: axial evolution of global Nusselt number. Discrete results and interpolating model.

bal elementary cell [equations (14) and (15)]. In addition we only introduce in equation (24) the well known axial evolution of the flow average temperatures for the (H2) condition ( $i = A, B$ ) [1]

$$\langle \Theta_{cm}^* \rangle_i = 4z_i^*. \quad (25)$$

The discrete axial profiles obtained from this oper-

ation are fitted satisfactorily by an equation in the form (see Fig. 13):

$$Nu_{H2,AB} = \frac{1}{A_H + B_H z_{AB}^*} + C_H (1000z_{AB}^*)^{-D_H} \exp(-30z_{AB}^*). \quad (26)$$

Equation (26) secures less than 1.5% error in cal-

culating the flow average global temperature ( $\langle \Theta_{\text{cm}}^* \rangle_{\text{AB}}$ ).

Like the ( $T$ )-condition, the relationship between global Nusselt number and geometry of the channel section is described by means of parameters  $A_{\text{H}}$ ,  $B_{\text{H}}$ ,  $C_{\text{H}}$ ,  $D_{\text{H}}$ , whose values calculated for the six selected  $DR$  values are adequately fitted by the following empirical expressions:

$$A_{\text{H}}(DR) = 0.1215 + 0.4114DR^{0.2451} \times \exp(-0.0477DR) \quad (27)$$

$$B_{\text{H}}(DR) = 1.5727DR^{-0.5550} \exp(-0.1807DR) \quad (28)$$

$$C_{\text{H}}(DR) = 9.2610 - 4.7589DR^{-0.0121} \times \exp(-0.2717DR) \quad (29)$$

$$D_{\text{H}}(DR) = 0.3889 - 0.0072DR^{-0.0338} \times \exp(-0.1975DR) \quad (30)$$

Notice that in the limit of  $z_{\text{AB}}^* \rightarrow \infty$  we always have  $Nu_{\text{H2,AB}}$  approaching zero, except for  $DR \rightarrow \infty$ . This peculiarity is due to the procedure for combination of the results of Sections A and B.

We recommend use of equations (17)–(22) and (26)–(30) only in the tested range of  $DR$  values (0.778–11.628).

## CONCLUSIONS

A generalized numerical solution of the forced convection heat transfer problem in the channels of plate-type monolith structures has been obtained for constant physical properties of the fluid stream and fully developed laminar flow. The analysis, which also involved the definition of the developed velocity profile, has been carried out using the finite element method (FEM) for numerical solution of the governing partial differential equations. Simulations were made with reference to a generalized two-channel geometric model derived from inspection of commercial plate-type monolith catalysts for SCR-DeNO<sub>x</sub> applications.

The results of the hydrodynamic problem have been summarized in a global average velocity profile given as a function of the single geometrical parameter  $DR$ .

The temperature problem has been solved both for a constant wall temperature ( $T$ -condition) and for a peripherally and axially constant heat flow ( $H2$ -condition). A global Nusselt number ( $Nu_{\text{AB}}$ ) has been derived in order to describe wall-gas heat transfer in the two-channel duct according to a simple one-dimensional approach. Also, equations have been derived to fit FEM results, representing the dependence of the axial evolution of the global Nusselt number ( $Nu_{\text{AB}}(z_{\text{AB}}^*)$ ) on the cross-sectional shape (i.e. on  $DR$ ). This has resulted in the definition of design expressions ( $Nu_{\text{T,AB}}(DR, z_{\text{AB}}^*)$  and  $Nu_{\text{H2,AB}}(DR, z_{\text{AB}}^*)$ ),

which allow calculation of the axial profile of the global bulk temperature along the monolith ducts by simple integration of the macroscopic one-dimensional energy balance for every configuration of the present geometrical model.

The values of the Nusselt number calculated in this work for the  $T$ - and the  $H2$ -boundary conditions represent asymptotic limits for the dimensionless gas-solid mass transfer coefficients (Sherwood numbers) in plate-type monolith catalysts. Application of the present results to design methods for chemical reactors using plate-type monoliths in the selective catalytic reduction of nitrogen oxides will be reported in a future paper.

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